



TOPICS COVERED IN THIS LECTURE

1. ISOPARAMETRIC QUADS AND HIGHER ORDER ELEMENTS IN 2D

2. BEAM EXAMPLES USING DIFFERENT 2D ELEMENTS

3. NUMERICAL INTEGRATION IN 1D AND 2D





ISOPARAMETRIC QUADS AND HIGHER ORDER ELEMENTS







$$u_{j}(\xi_{1},\xi_{2}) = \sum_{I=1}^{4} N_{I}(\xi_{1},\xi_{2}) U_{j}^{I} \qquad N_{I}(\xi_{1}^{J},\xi_{2}^{J}) = \delta_{IJ}; \ (I,J=1,\dots4)$$

$$(\xi_{1}^{J},\xi_{2}^{J}) = (I,J) \qquad (\xi_{2}^{J},\xi_{3}^{J}) = (I,J)$$







$$N_I(\xi_1, \xi_2) = \frac{1}{4} (1 + \xi_1^I \xi_1) (1 + \xi_2^I \xi_2)$$

$$\mathbf{J} \equiv \left[\frac{\partial x_j}{\partial \xi_i}\right] = \left[\sum_{I=1}^4 \frac{\partial N_I}{\partial \xi_i} (\xi_1, \xi_2) x_j^I\right]$$

Shape functions $N_I(\xi_1, \xi_2)$ and coordinate transformation matrix **J** for 4-node isoparametric quads

$$\mathbf{J} = \begin{bmatrix} \frac{1}{4} \sum_{I=1}^{4} \xi_{1}^{I} (1+\xi_{2}^{I}\xi_{2}) x_{1}^{I} & \frac{1}{4} \sum_{I=1}^{4} \xi_{1}^{I} (1+\xi_{2}^{I}\xi_{2}) x_{2}^{I} \\ \\ \frac{1}{4} \sum_{I=1}^{4} \xi_{2}^{I} (1+\xi_{1}^{I}\xi_{1}) x_{1}^{I} & \frac{1}{4} \sum_{I=1}^{4} \xi_{2}^{I} (1+\xi_{1}^{I}\xi_{1}) x_{2}^{I} \end{bmatrix}$$





$$\boldsymbol{\epsilon} = \begin{bmatrix} \frac{\partial u_1}{\partial x_1} \\ \frac{\partial u_2}{\partial x_2} \\ \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \end{bmatrix} = \begin{bmatrix} \frac{\partial \xi_1}{\partial x_1} & \frac{\partial \xi_2}{\partial x_1} & 0 & 0 \\ 0 & 0 & \frac{\partial \xi_1}{\partial x_2} & \frac{\partial \xi_2}{\partial x_2} \\ \frac{\partial \xi_1}{\partial x_2} & \frac{\partial \xi_2}{\partial x_2} & \frac{\partial \xi_1}{\partial x_1} & \frac{\partial \xi_2}{\partial x_1} \end{bmatrix} \begin{bmatrix} \frac{\partial u_1}{\partial \xi_2} \\ \frac{\partial u_2}{\partial \xi_1} \\ \frac{\partial u_2}{\partial \xi_2} \end{bmatrix}$$

Recall definition of matrix $\mathbf{A} = \frac{\mathbf{1}}{\det \mathbf{J}} \begin{bmatrix} J_{22} & -J_{12} \mathbf{A} & 0 & 0 \\ 0 & 0 & -J_{21} & J_{11} \\ -J_{21} & J_{11} & J_{22} & -J_{12} \end{bmatrix}$











$$\mathcal{P}_{int}^{e} = \int_{A_{e}} \frac{1}{2} [\boldsymbol{\epsilon}^{T} \boldsymbol{\sigma}(x_{1}, x_{2})] dA = \frac{1}{2} \mathbf{q}_{e}^{T} \left[\int_{\boldsymbol{\xi}} [\mathbf{G}^{T} \mathbf{A}^{T} \mathbf{L} \mathbf{A} \mathbf{G}] \det(\mathbf{J}) d\boldsymbol{\xi} \right] \mathbf{q}_{e}$$
$$= \frac{1}{2} \mathbf{q}_{e}^{T} \mathbf{k}_{e} \mathbf{q}_{e}$$
Element stiffness matrix: \mathbf{k}_{e}

$$-\mathcal{P}_{ext}^{e} = \int_{A_{e}} [\mathbf{u}^{T}(x_{1}, x_{2})\mathbf{b}(x_{1}, x_{2})] dA + \int_{\partial A_{e}} [\mathbf{u}^{T}(x_{1}, x_{2})\mathbf{t}(x_{1}, x_{2})] dl$$
$$= \mathbf{q}_{e}^{T} \left[\int_{\boldsymbol{\xi}} [\mathbf{N}^{T} \mathbf{b}] \det(\mathbf{J}) d\boldsymbol{\xi} + \int_{\partial \boldsymbol{\xi}} [\mathbf{N}^{T} \mathbf{t}] dl(\boldsymbol{\xi}) \right]$$
$$= \mathbf{q}_{e}^{T} \mathbf{f}_{e} \qquad \text{Element force vector: } \mathbf{f}_{e}$$









FEM IN 2D: ISOPARAMETRIC 9-NODE QUAD ELEMENTS





Recall quadratic Lagrangian functions $L_i(\xi)$ in interval [-1,1]

$$x_j\left(\xi_1^I,\xi_2^I\right) = x_j^I$$

Shape functions are products: $N_I(\xi_1, \xi_2) = L_i(\xi_i) L_j(\xi_2)$

$$N_I(\xi_1^J, \xi_2^J) = \delta_{IJ}; \quad (I, J = 1, \dots 9)$$





$N_{4}(\xi_{1},\xi_{2}) = L_{1}(\xi_{1}) L_{3}(\xi_{2}) \qquad N_{7}(\xi_{1},\xi_{2}) = L_{2}(\xi_{1}) L_{3}(\xi_{2}) \qquad N_{3}(\xi_{1},\xi_{2}) = L_{3}(\xi_{1}) L_{3}(\xi_{2})$



 $N_{1}(\xi_{1},\xi_{2}) = L_{1}(\xi_{1}) L_{1}(\xi_{2}) \qquad N_{5}(\xi_{1},\xi_{2}) = L_{2}(\xi_{1}) L_{1}(\xi_{2}) \qquad N_{2}(\xi_{1},\xi_{2}) = L_{3}(\xi_{1}) L_{1}(\xi_{2})$

Disadvantage of 9-node quad elements: needless increase of bandwidth







Way to eliminate internal nodes: static condensation



Change element stiffness & force of boundary nodes by:

$$egin{array}{rcl} \mathbf{k}^e_{ij} &
ightarrow \mathbf{k}^e_{ij} - \mathbf{k}^e_{i9} [\mathbf{k}^e_{99}]^{-1} \mathbf{k}^e_{9j} \ \mathbf{f}^e_i &
ightarrow \mathbf{f}^e_i - \mathbf{k}^e_{i9} [\mathbf{k}^e_{99}]^{-1} \mathbf{f}^e_9 \end{array}$$

$\left\lceil \mathbf{K}_{11}^{G} \right.$	\mathbf{K}_{12}^G	\mathbf{K}_{13}^G	\mathbf{K}_{14}^G	\mathbf{K}_{15}^{G}	\mathbf{K}_{16}^{G}	\mathbf{K}_{17}^G	\mathbf{K}_{18}^G	\mathbf{k}_{19}^e	$\left\lceil \mathbf{Q}_{1} ight ceil$		$\left[\mathbf{F}_{1}^{G}\right]$	
\mathbf{K}_{21}^G	\mathbf{K}_{22}^{G}	\mathbf{K}_{23}^G	\mathbf{K}_{24}^G	\mathbf{K}_{25}^{G}	\mathbf{K}_{26}^G	\mathbf{K}_{27}^G	\mathbf{K}_{28}^G	\mathbf{k}^e_{29}	\mathbf{Q}_2		\mathbf{F}_2^G	
\mathbf{K}_{31}^G	\mathbf{K}_{32}^G	\mathbf{K}_{33}^G	\mathbf{K}_{34}^G	\mathbf{K}_{35}^G	\mathbf{K}_{36}^G	\mathbf{K}_{37}^G	\mathbf{K}_{38}^G	\mathbf{k}^e_{39}	\mathbf{Q}_3		\mathbf{F}_3^G	
\mathbf{K}_{41}^G	\mathbf{K}^G_{42}	\mathbf{K}^G_{43}	\mathbf{K}_{44}^G	\mathbf{K}^G_{45}	\mathbf{K}_{46}^G	\mathbf{K}^G_{47}	\mathbf{K}^G_{48}	\mathbf{k}^e_{39}	\mathbf{Q}_4		\mathbf{F}_4^G	
\mathbf{K}_{51}^G	\mathbf{K}_{52}^{G}	\mathbf{K}_{53}^G	\mathbf{K}_{54}^G	\mathbf{K}_{55}^{G}	\mathbf{K}_{56}^{G}	\mathbf{K}_{57}^G	\mathbf{K}_{58}^G	\mathbf{k}^e_{59}	\mathbf{Q}_5	=	\mathbf{F}_5^G	
\mathbf{K}_{61}^G	\mathbf{K}_{62}^G	\mathbf{K}_{63}^G	\mathbf{K}_{64}^G	\mathbf{K}_{65}^{G}	\mathbf{K}_{66}^{G}	\mathbf{K}_{67}^G	\mathbf{K}_{68}^G	\mathbf{k}^e_{69}	\mathbf{Q}_{6}		\mathbf{F}_6^G	
\mathbf{K}_{71}^G	\mathbf{K}_{72}^{G}	\mathbf{K}_{73}^G	\mathbf{K}_{74}^G	\mathbf{K}_{75}^{G}	\mathbf{K}_{76}^{G}	\mathbf{K}_{77}^G	\mathbf{K}_{78}^G	\mathbf{k}^e_{79}	\mathbf{Q}_7		\mathbf{F}_7^G	
\mathbf{K}^G_{81}	\mathbf{K}^G_{82}	\mathbf{K}^G_{83}	\mathbf{K}^G_{84}	\mathbf{K}^G_{85}	\mathbf{K}^G_{86}	\mathbf{K}^G_{87}	\mathbf{K}^G_{88}	\mathbf{k}^e_{89}	\mathbf{Q}_8		\mathbf{F}_8^G	
\mathbf{k}^{e}_{91}	\mathbf{k}_{92}^e	\mathbf{k}_{93}^e	\mathbf{k}_{94}^e	\mathbf{k}^e_{95}	\mathbf{k}^e_{96}	\mathbf{k}^e_{97}	\mathbf{k}^e_{98}	\mathbf{k}^{e}_{99}	\mathbf{Q}_{9}		$\left[\mathbf{f}_{9}^{e} ight]$	

Equilibrium equation for node 9 solved immediately











Shape functions of node I are products of line equations with remaining nodes

$$N_1(\xi_1,\xi_2) = -\frac{1}{4}(1-\xi_1)(1-\xi_2)(1+\xi_1+\xi_2), \ N_5(\xi_1,\xi_2) = \frac{1}{2}(1-\xi_1^2)(1-\xi_2)$$

$$N_2(\xi_1,\xi_2) = -\frac{1}{4}(1+\xi_1)(1-\xi_2)(1-\xi_1+\xi_2), \ N_6(\xi_1,\xi_2) = \frac{1}{2}(1+\xi_1)(1-\xi_2^2)$$

$$N_3(\xi_1,\xi_2) = -\frac{1}{4}(1+\xi_1)(1+\xi_2)(1-\xi_1-\xi_2), \ N_7(\xi_1,\xi_2) = \frac{1}{2}(1-\xi_1^2)(1+\xi_2)$$

$$N_4(\xi_1,\xi_2) = -\frac{1}{4}(1-\xi_1)(1+\xi_2)(1+\xi_1-\xi_2), \ N_8(\xi_1,\xi_2) = \frac{1}{2}(1-\xi_1)(1-\xi_2^2)$$





Shape functions of node I are products of equations avoiding that node



Triangular coordinates satisfy: $\xi_1 + \xi_2 + \xi_3 = 1$ $N_6 = 4 \xi_3 \xi_1$





BEAM EXAMPLES USING DIFFERENT 2D ELEMENTS







Problem will be solved using different elements in 2D

Analytical solution found with the help of Airy's stress function satisfies average equilibrium conditions at each end, while FEM solution considers correct boundary conditions (all nodes fixed at left, have forces applied at right)

Analytical solution correct away from boundaries (boundary layers)





FEM solution of cantilever beam showing end effects

Step: Step-1 Frame: 100 Total Time: 1.000000





 Y ODB: N-CPE4.adb Abaque/Standard 6.12-3 Wed Oct D1 17: 26: 56 GMT+D2: DD 2014
 Step: Step-1
 X Increment 100: Step Time = 1.000 Primary Var: 5, 511 Deformed Var: U Deformation Scale Factor: +1.000e+D0





Airy stress function has correct axial, shear forces and moment at ends

$$\phi(x_1, x_2) = \frac{h^2}{4} \frac{F}{L} \frac{x_2}{h} \left(\left(\frac{x_1}{L} - 1\right) - 3\frac{x_1}{L} \left(\frac{x_2}{h}\right)^2 \right) \left(\frac{L}{h}\right)^2$$

Left:
$$(x_1=0)$$

 $N=0$
 $\sigma_{11} = \frac{\partial^2 \phi}{\partial x_2 \partial x_2} = \frac{3}{2} \frac{x_2}{h} \frac{F}{L} \left(\frac{x_1}{L} - 1\right) \left(\frac{L}{h}\right)^2$
 $N=0$
 $N=0$

$$V = I$$

$$\sigma_{22} = \frac{\partial^2 \phi}{\partial x_1 \partial x_1} = 0$$

$$V = I$$

$$V = I$$

M = 0

$$\sigma_{12} = -\frac{\partial^2 \phi}{\partial x_1 \partial x_2} = -\frac{3}{4} \frac{F}{L} \left(\left(\frac{x_2}{h}\right)^2 - 1 \right) \frac{L}{h}$$





FEM solution using 4 4-node quads







FEM solution using 16 4-node quads







FEM solution using 64 4-node quads







FEM solution using 4 8-node quads



























Beam is simply supported at each end. Due to symmetry only half needs to be modeled.

Analytical solution found with the help of Airy's stress function satisfies average equilibrium conditions at each end, while FEM solution considers correct boundary conditions



FEM solution of simply supported beam showing end effects





ODB: N-CPE4.adb Abagus/Standard 6.12-3 Wed Oct D1 17: 35: 42 GMT+D2: 00 2014

Step: Step-1

Primary Var: 5, 511

Inciement 100: Step Time = 1.000

Deformed Var: U Deformation Scale Factor: +1.000e+00

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$$\phi(x_1, x_2) = \frac{ph^2}{40} \left\{ \left(\frac{x_2}{h}\right)^3 \left(2 - \left(\frac{x_2}{h}\right)^2\right) + 5\left(\frac{L}{h}\right)^2 \left[\left(\frac{x_1}{L}\right)^2 \left(\left(\frac{x_2}{h}\right)^3 - 3\left(\frac{x_2}{h}\right) - 2\right) - \left(\frac{x_2}{h}\right)^3\right] \right\}$$

$$\sigma_{11} = \frac{\partial^2 \phi}{\partial x_2 \partial x_2} = \frac{3p}{4} \left(\frac{L}{h}\right)^2 \frac{x_2}{h} \left[\left(\frac{x_1}{L}\right)^2 - 1 \right] + \frac{p}{10} \frac{x_2}{h} \left(3 - 5 \left(\frac{x_2}{h}\right)^2 \right)$$

$$\sigma_{22} = \frac{\partial^2 \phi}{\partial x_1 \partial x_1} = \frac{p}{4} \left[\left(\frac{x_2}{h}\right)^3 - 3\left(\frac{x_2}{h}\right) - 2 \right]$$

$$\sigma_{12} = -\frac{\partial^2 \phi}{\partial x_1 \partial x_2} = \frac{3p}{4} \left(\frac{L}{h}\right) \left(\frac{x_1}{L}\right) \left[1 - \left(\frac{x_2}{h}\right)^2\right]$$





FEM solution using 4 4-node quads







FEM solution using 16 4-node quads







FEM solution using 64 4-node quads







FEM solution using 4 8-node quads



















NUMERICAL INTEGRATION IN 1D AND 2D



NUMERICAL INTEGRATION IN 1D ELEMENTS









Gaussian quadrature of 1, 2 and 3 points 1 point Gauss : $n_I = 1$; $\xi^1 = 0$ $w_1 = 2$; $R = \frac{1}{3} \frac{d^2 f}{d\xi^2}(\xi^*)$; $\xi^* \in (-1, 1)$ 2 point Gauss : $n_I = 2$; $\xi^1 = -\frac{1}{\sqrt{3}}, \ \xi^2 = \frac{1}{\sqrt{3}}$

$$w_1 = w_2 = 1; \quad R = \frac{1}{135} \frac{d^4 f}{d\xi^4}(\xi^*)$$

3 point Gauss :
$$n_I = 3$$
; $\xi^1 = -\frac{\sqrt{3}}{\sqrt{5}}, \ \xi_2 = 0, \ \xi^3 = \frac{\sqrt{3}}{\sqrt{5}}$
 $w_1 = w_3 = \frac{5}{9}, \ w_2 = \frac{8}{9}; \ R = \frac{1}{15750} \frac{d^6 f}{d\xi^6}(\xi^*)$





General Gaussian quadrature of n_I points is $2n_I$ accurate

Gauss points : ξ^{I} are roots of Legendre polynomial $P_{n_{I}}(\xi^{I}) = 0$

weight :
$$w_I = \frac{2}{\left[(1 - (\xi^I)^2)\right] \left[\frac{dP_{n_I}}{d\xi}(\xi^I)\right]^2}$$

remainder :
$$R = \frac{2^{(2n_I)}(n_I!)^4}{(2n_I+1)[(2n_I)!]^3} \frac{d^{(2n_I)}f}{d\xi^{(2n_I)}}(\xi^*)$$

$$n_{I}$$
 order Legendre : $P_{n_{I}}(\xi) \equiv \frac{1}{2^{n_{I}}(n_{I})!} \frac{d^{(n_{I})}}{d\xi^{(n_{I})}} (\xi^{2} - 1)^{n_{I}}$





General Gaussian quadrature in 2D uses master element

$$\int_{-1}^{+1} \int_{-1}^{+1} f(\xi_1, \xi_2) \, d\xi_1 d\xi_2 = \sum_{I=1}^{I=n_I} \sum_{J=1}^{J=n_I} w_I w_J f(\xi_1^I, \xi_2^J)$$



A 2×2 Gauss integration uses grid points with coordinates and weights taken from 1D

 $(\xi_1^{\ l},\xi_2^{\ l}) = (-1/\sqrt{3}, -1/\sqrt{3}) \quad (\xi_1^{\ l},\xi_2^{\ 2}) = (-1/\sqrt{3}, 1/\sqrt{3})$

 ξ_1